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A Discrete Ordinates Algorithm for Domains with Embedded Boundaries

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Abstract

Embedded boundary methods model fluid flows in complex geometries by treating boundaries as tracked interfaces in a regular mesh. Though often referred to as "Cartesian grid" methods, they are equally well-suited to axisymmetric problems. This paper describes a formulation of the discrete ordinates method for radiative transfer calculations with embedded boundaries. The method uses diamond-difference stencils in the interior with a conservative extension to boundary cells based on a volume-of-fluid approach. Numerical examples are presented in both 2D Cartesian and axisymmetric geometries, including a model of the BERL 300kW natural gas burner.

Nomenclature

μ, ξ, η	Direction cosines
ω	Angle of revolution about ξ axis, $\tan^{-1}(\eta/\mu)$
p,q	Ordinate indices, axisymmetric coordinates
m	Ordinate index, Cartesian coordinates
$w_{p,q}, w_m$	Ordinate weight
$\Omega_{p,q}, \Omega_m$	Ordinate direction unit vector
$I_{p,q}, I_m$	Radiant intensity, $W/(m^2 \cdot Sr)$
I_b	Blackbody intensity, $\sigma_b T^4/\pi$, W/(m ² · Sr)
κ , σ	Absorption and scattering coefficients, m ⁻¹
ϵ, ρ	Wall emissivity and reflectivity
S	Source term, W/(m ³ · Sr)
α	Dimensionless parameter in angular difference
$oldsymbol{w}$	Superscript for quantities at the wall
L^{w}	Length of embedded boundary segment, m
$\hat{m{n}}$	Unit normal to embedded boundary segment
$oldsymbol{F}$	Volume fraction, $\in [0, 1]$
f .	Area fraction, $\in [0,1]$
$\lceil x \rceil$	$\max(x,0)$
x	$\min(x,0)$

Introduction

Embedded boundary methods have proven to be a robust and competitive approach to computing fluid flows in complex geometries. First introduced in [1], they have been used successfully both in compressible calculations with adaptive mesh refinement [2], [3] and, more recently, in incompressible calculations as well [4]. In [5] we presented some results using this approach to model combustion in a gas-fired burner, with the radiative transfer effects discretized using a discrete ordinate method. Our intent in this paper is to give a more complete and validated presentation of the discrete ordinate method with embedded boundaries, which in [5] was only briefly introduced.

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The basic idea behind embedded boundaries is to use a regular mesh to cover the computational domain, then block off those parts of the grid that are inside the chamber walls or any solid obstructions. Previous work of this nature with the discrete ordinate method has been presented by Chai, et al. in [6]. In their approach computational cells are marked as being either fluid cells or body cells, giving the embedded boundary a "stairstep" appearance. In contrast, our present work treats the boundary more as a piecewise-linear tracked front, so a third kind of cell, the mixed cell, has been added to the model. This approach is consistent with that taken in [3], permitting us to use the same data structures for fluid dynamics and for radiation and to compute coupled solutions such as the ones in [5].

We will begin by briefly presenting the discrete ordinate discretization for an axisymmetric coordinate system. This is not new work, but will serve to introduce the notation and provide a framework for introducing the embedded boundary. The discretization for Cartesian coordinates is a limiting case of the axisymmetric discretization, and need not be presented separately. Next, we will develop stencils for representing the embedded boundary in Cartesian coordinates, and then extend these to cover the axisymmetric case. Finally, we will present validation results and some numerical examples.

Method with Flat Walls

The discrete ordinates representation of the radiative transport equation with isotropic scattering in axisymmetric coordinates is

$$\frac{\mu_{p,q}}{r} \frac{\partial (rI_{p,q})}{\partial r} + \xi_p \frac{\partial I_{p,q}}{\partial z} - \frac{1}{r} \frac{\partial (\eta I)}{\partial \omega} + (\kappa + \sigma)I_{p,q} = \kappa I_b + \frac{\sigma}{4\pi} \sum_{p',q'} w_{p',q'} I_{p',q'}, \tag{1}$$

where the ordinate weights $w_{p,q}$ are normalized so that $\sum_{p,q} w_{p,q} = 4\pi$. The subscript p identifies a level of ordinate directions with a particular value of ξ , while q indexes the separate ordinates sharing that level. The angular term involving η and ω is written without subscripts because it represents an interaction between ordinate directions; this term will not be discretized directly, rather it will be used to maintain joint conservation with the $\partial/\partial r$ term. The right hand side of (1), representing the emission and scattering sources, respectively, we will abbreviate as S.

We discretize (1) as a conservation relation for each cell,

$$\frac{\mu_{p,q}}{r_{i}\Delta r} (r_{i+1/2}I_{p,q,i+1/2,j} - r_{i-1/2}I_{p,q,i-1/2,j})
+ \frac{\xi_{p}}{\Delta z} (I_{p,q,i,j+1/2} - I_{p,q,i,j-1/2})
+ \frac{1}{r_{i}w_{p,q}} (\alpha_{p,q+1/2}I_{p,q+1/2,i,j} - \alpha_{p,q-1/2}I_{p,q-1/2,i,j}) + (\kappa + \sigma)I_{p,q,i,j} = S_{i,j},$$
(2)

where quantities with half-indices exist at cell faces. The α term represents propagation from one ordinate direction to the next due to curvature of the mesh; a uniform isotropic radiation field with no emission, absorption or scattering yields the relation

$$\alpha_{p,q+\frac{1}{2}} - \alpha_{p,q-\frac{1}{2}} = -w_{p,q}\mu_{p,q}. \tag{3}$$

Detailed derivations of these equations can be found in [7] and [8], as can most aspects of the solution procedure. To summarize briefly, we relate the face- and cell-based quantities in (2) using diamond-difference formulas,

$$I_{p,q,i-1/2,j} + I_{p,q,i+1/2,j} = I_{p,q,i,j-1/2} + I_{p,q,i,j+1/2} = I_{p,q-1/2,i,j} + I_{p,q+1/2,i,j} = 2I_{p,q,i,j}.$$
(4)

For each ordinate direction we sweep through the mesh in the direction of propagation. When μ and ξ are positive this involves solving for $I_{p,q,i,j}$, $I_{p,q,i+\frac{1}{2},j}$, $I_{p,q,i,j+\frac{1}{2}}$ and $I_{p,q+\frac{1}{2},i,j}$ in terms of $I_{p,q,i-\frac{1}{2},j}$, $I_{p,q,i,j-\frac{1}{2}}$ and $I_{p,q-\frac{1}{2},i,j}$. In some cases one of the computed edge quantities can be negative. Since negative fluxes are nonphysical and can lead to spurious oscillations, we set the offending quantities to zero and recompute the other dependent fluxes. The ability to perform this flux limiting in a relatively straightforward manner is

the main motivation for choosing the cell- and face-based diamond difference formulation over the otherwise equivalent node-based bilinear formulation.

To incorporate the flux limiting operation into a vector loop, we observe that the cell-centered flux $I_{p,q,i,j}$ takes on its smallest possible value when the correct combination of dependent face fluxes is limited. The sign of one flux may depend on which others are limited, so it is necessary to consider all eight combinations separately. The additional computation is not excessive, however, since many terms common to all flux combinations may be precomputed. The general form for the central flux computation is

$$I_{p,q,i,j} = \frac{S_{i,j} + \frac{\mu_{p,q}}{r_i \Delta r} (\ell_{\mu} r_{i+1/2} + r_{i-1/2}) I_{p,q,i-1/2,j} + \frac{\xi_p}{\Delta z} (\ell_{\xi} + 1) I_{p,q,i,j-1/2} + \frac{1}{r_i w_{p,q}} (\ell_{\alpha} \alpha_{p,q+1/2} + \alpha_{p,q-1/2}) I_{p,q-1/2,i,j}}{\kappa + \sigma + \ell_{\mu} \frac{2\mu_{p,q} r_{i+1/2}}{r_i \Delta r} + \ell_{\xi} \frac{2\xi_p}{\Delta z} + \ell_{\alpha} \frac{2\alpha_{p,q+1/2}}{r_i w_{p,q}}}$$
(5

where each of ℓ_{μ} , ℓ_{ξ} and ℓ_{α} is 0 or 1 depending on whether the corresponding face is or is not limited, respectively. Once $I_{p,q,i,j}$ has been determined as the minimum of its eight possible values, we set

$$I_{p,q,i+1/2,j} = \max(2I_{p,q,i,j} - I_{p,q,i-1/2,j}, 0)$$
(6)

and likewise with $I_{p,q,i,j+\frac{1}{2}}$ and $I_{p,q+\frac{1}{2},i,j}$.

The computation begins at the outer boundary with inward-directed ordinates and proceeds through increasing values of $\mu_{p,q}$ for each ξ_p . Zero-weight starting directions are used to obtain each I_{p,γ_2} . These starting directions are additional ordinates added to the set, one for each p, with $\mu_{p,0} = -\sqrt{1-\xi_p^2}$. With no angular extent, they have $I_{p,-\gamma_2,i,j} = I_{p,0,i,j} = I_{p,\gamma_2,i,j}$ and their discretization reduces to the one for Cartesian coordinates:

$$\frac{\mu_{p,0}}{\Delta r}(I_{p,0,i+1/2,j} - I_{p,0,i-1/2,j}) + \frac{\xi_p}{\Delta z}(I_{p,0,i,j+1/2} - I_{p,0,i,j-1/2}) + (\kappa + \sigma)I_{p,0,i,j} = S_{i,j}.$$
 (7)

At the r=0 boundary we follow the suggestion of [8] and set each outward $I_{p,q}$ to the w-weighted average of the inward fluxes on the same p-level:

$$I_{p,q} = \frac{\sum_{\mu_{p,q'} < 0} w_{p,q'} I_{p,q'}}{\sum_{\mu_{p,q'} < 0} w_{p,q'}}, \quad \mu_{p,q} > 0.$$
(8)

Reflecting wall boundaries and the scattering source can be accounted for by iteration in the manner of [9]. Since our primary focus in this paper is the treatment of wall boundaries, we will not concern ourselves here with the details of the scattering solution. The equation to be satisfied at the "outer" (r) wall of the domain is

$$I_{p,q} = \epsilon I_b^w + \frac{\rho}{\pi} \sum_{\mu_{p',q'} > 0} w_{p',q'} \mu_{p',q'} I_{p',q'}, \quad \mu_{p,q} < 0, \tag{9}$$

and similarly for the "top" and "bottom" (z) walls. This equation implicitly assumes that the ordinate set has a half-range first moment exactly equal to π . Ordinate sets with this property—for walls aligned with the coordinate axes—are given in [10] and [11].

It is not necessary to store interior fluxes for all ordinate directions at once, but it is convenient to store fluxes for each ordinate along edges where the radiation meets a wall. This permits us to update the reflection sources after each ordinate sweep, so that all of the source information is up-to-date before starting the sweep for the next ordinate direction.

The radiation solver couples back into the equation for fluid energy conservation through the quantity

$$\nabla \cdot q_{rad} = 4\pi \kappa I_b - \kappa \sum_{p,q} w_{p,q} I_{p,q}. \tag{10}$$

Discretization at an Embedded Boundary

Many calculations involve more complicated geometries, which we model as embedded boundaries. (This is often called the "Cartesian Grid" method, even though it works just as well for axisymmetric coordinates.)

Geometry information is represented on the entire grid as volume fractions for each cell and area fractions for each face. These fractions are all 1 for fluid cells and all 0 for body cells, with intermediate values in the partial cells at the boundary.

To adapt the radiation solver to this algorithm framework we derived a form of the discrete ordinate discretization which is valid for all three cell types and which reduces to the usual form given above for cells entirely in the fluid. The advantages of this approach are that separate coding for boundary cells is minimized, boundary cells are updated in the same vector loop as interior cells, and they are updated in the proper order following the radiation flow so that information lags do not degrade the convergence rate.

Cartesian Coordinates

It is easiest to start the derivation with a true Cartesian coordinate system, and since this formulation applies to the zero-weight starting directions for axisymmetric coordinates the formulas will be directly useful. The conservation relation in each cell becomes

$$\frac{\mu_{m}}{\Delta x} (f_{i+1/2,j} I_{m,i+1/2,j} - f_{i-1/2,j} I_{m,i-1/2,j}) + \frac{\xi_{m}}{\Delta y} (f_{i,j+1/2} I_{m,i,j+1/2} - f_{i,j-1/2} I_{m,i,j-1/2}) + \frac{\hat{n} \cdot \Omega_{m}}{\Delta x \Delta y} L_{i,j}^{w} I_{m,i,j}^{w} + (\kappa + \sigma) F_{i,j} I_{m,i,j} = F_{i,j} S_{i,j}, \tag{11}$$

where F and f are volume and area fractions, respectively, and $I_{m,i,j}^w$ is the intensity at the wall. (We use a single subscript, m, to denote the ordinate direction in Cartesian coordinates, since there is no angular coupling and hence no need to distinguish the sets of ordinates sharing a value of ξ .) The contribution from flux across the embedded boundary appears to require the unit normal \hat{n} and the wall segment length L^w for each cell, but this term can be rewritten in terms of volume and area fractions by considering a uniform flow in empty space:

$$\frac{\mu_m}{\Delta x} (f_{i+1/2,j} - f_{i-1/2,j}) + \frac{\xi_m}{\Delta y} (f_{i,j+1/2} - f_{i,j-1/2}) + \frac{\hat{n} \cdot \Omega_m}{\Delta x \Delta y} L_{i,j}^w = 0.$$
 (12)

The diamond difference formulation is based on taking a multilinear profile across each cell in order to express $I_{m,i+1/2,j}$ in terms of $I_{m,i-1/2,j}$ and $I_{m,i,j}$, and similarly for $I_{m,i,j+1/2}$. With embedded boundaries we must consider the possibility that $I_{m,i-1/2,j}$ may not be available, if that face of the cell is entirely outside the fluid region. Permitting the discretization to drop to first order at the boundary, we modify the diamond difference formulas as follows:

$$I_{m,i+1/2,j} = I_{m,i,j} + f_{i-1/2,j} (I_{m,i,j} - I_{m,i-1/2,j})$$
(13)

$$I_{m,i,j+1/2} = I_{m,i,j} + f_{i,j-1/2}(I_{m,i,j} - I_{m,i,j-1/2}).$$
(14)

(These formulas change in the obvious way for rays traveling in the $-\mu$ and $-\xi$ directions.)

If $\hat{n} \cdot \Omega_m < 0$, the $I^w_{m,i,j}$ term represents a source to the cell from emission and reflection at the boundary—we precompute these sources $S^w_{m,i,j}$ for all boundary cells before beginning the transport sweep for each ordinate direction. If $\hat{n} \cdot \Omega_m > 0$, this term is flux from the cell to the boundary and becomes part of the cell update. We use $I^w_{m,i,j} = I_{m,i,j}$ to compute these fluxes since the relation is already first order at the boundary.

Using the modified profile across the cell, the conservation relation becomes

$$\frac{\mu_{m}}{\Delta x} [f_{i+1/2,j}(1+f_{i-1/2,j})I_{m,i,j} - f_{i-1/2,j}(1+f_{i+1/2,j})I_{m,i-1/2,j}] + \frac{\xi_{m}}{\Delta y} [f_{i,j+1/2}(1+f_{i,j-1/2})I_{m,i,j} - f_{i,j-1/2}(1+f_{i,j+1/2})I_{m,i,j-1/2}] + \left[\frac{\hat{n} \cdot \Omega_{m}}{\Delta x \Delta y} L_{i,j}^{w}\right] I_{m,i,j} + (\kappa + \sigma) F_{i,j} I_{m,i,j} = F_{i,j} S_{i,j} + S_{m,i,j}^{w}, \tag{15}$$

where $\lceil x \rceil = \max(x,0)$. This expression is solved for $I_{m,i,j}$, and modified as before to limit the fluxes at the extrapolated faces. Note that since $I_{m,i,j}^w$ is not extrapolated, it does not have to be limited.

For each ordinate direction, after completing the transport sweep the reflection source is updated for all boundary cells with $\hat{n} \cdot \Omega_m > 0$. This update requires that we store intensities for all ordinate directions in the boundary cells, though we still do not have to store more than one direction at a time in the interior. (For storage of boundary information we use a sparse data structure, which is essentially a list of mixed cells with pointers into the main grid.)

The general form of the boundary condition is

$$S_{m,i,j}^{w} = -\left[\frac{\hat{n} \cdot \Omega_{m}}{\Delta x \Delta y} L_{i,j}^{w}\right] \left\{ \epsilon I_{b}^{w} + \rho \frac{\sum_{\hat{n} \cdot \Omega_{m'} > 0} (\hat{n} \cdot \Omega_{m'}) w_{m'} L_{i,j}^{w} I_{m',i,j}^{w}}{\sum_{\hat{n} \cdot \Omega_{m'} > 0} (\hat{n} \cdot \Omega_{m'}) w_{m'} L_{i,j}^{w}} \right\},$$
(16)

where $\lfloor x \rfloor = \min(x,0)$. To apply this in practice we precompute the denomenator of the reflection term, and update the numerator after each ordinate sweep. If the ordinate set had the correct half-range first moment in the direction \hat{n} the denomenator would become simply $\pi L^w_{i,j}$ (cf. equation (9)), but it is not possible to choose an ordinate set for which $\sum_{\hat{n}:\Omega_{-}>0} (\hat{n} \cdot \Omega_m) w_m = \pi$ for all possible \hat{n} .

Cylindrical Coordinates

There are no new difficulties involved in taking this embedded boundary method to an axisymmetric coordinate system, the formulas just become more complicated. The fractions $F_{i,j}$ and $f_{i,j\pm\frac{1}{2}}$ are r-weighted so as to be true volume and area fractions in three dimensions. That is, the volumes and areas are derived for the annular regions which result when cells are rotated about the central axis. The cell conservation relation with volume and area fractions takes the form

$$\frac{\mu_{p,q}}{r_{i}\Delta r} (r_{i+1/2}f_{i+1/2,j}I_{p,q,i+1/2,j} - r_{i-1/2}f_{i-1/2,j}I_{p,q,i-1/2,j}) + \frac{\xi_{p}}{\Delta z} (f_{i,j+1/2}I_{p,q,i,j+1/2} - f_{i,j-1/2}I_{p,q,i,j-1/2}) + \frac{F_{i,j}}{r_{i}w_{p,q}} (\alpha_{p,q+1/2}I_{p,q+1/2,i,j} - \alpha_{p,q-1/2}I_{p,q-1/2,i,j}) + \frac{\hat{n} \cdot \Omega_{p,q}}{r_{i}\Delta r \Delta z} r_{i,j}^{w} L_{i,j}^{w} I_{p,q,i,j}^{w} + (\kappa + \sigma)F_{i,j}I_{p,q,i,j} = F_{i,j}S_{i,j}. \tag{17}$$

Uniform isotropic flow then yields an equation for the wall factor:

$$\frac{\mu_{p,q}}{r_{i}\Delta r}(r_{i+1/2}f_{i+1/2,j} - r_{i-1/2}f_{i-1/2,j}) + \frac{\xi_{p}}{\Delta z}(f_{i,j+1/2} - f_{i,j-1/2}) + \frac{F_{i,j}}{r_{i}w_{p,q}}(\alpha_{p,q+1/2} - \alpha_{p,q-1/2}) + \frac{\hat{n} \cdot \Omega_{p,q}}{r_{i}\Delta r\Delta z}r_{i,j}^{w}L_{i,j}^{w} = 0.$$
(18)

To see that the weighting on the angular term is correct, consider replacing this term with a source function of the form $S = \mu/r$.

The spatial differences can use the same modified diamond difference as in Cartesian coordinates, while no modification is necessary in the angular term. Eliminating extrapolated values from the conservation relation then yields

$$\begin{split} \frac{\mu_{p,q}}{r_{i}\Delta\tau} [r_{i+1/2}f_{i+1/2,j}(1+f_{i-1/2,j})I_{p,q,i,j} - f_{i-1/2,j}(r_{i-1/2}+r_{i+1/2}f_{i+1/2,j})I_{p,q,i-1/2,j}] + \\ \frac{\xi_{p}}{\Delta z} [f_{i,j+1/2}(1+f_{i,j-1/2})I_{p,q,i,j} - f_{i,j-1/2}(1+f_{i,j+1/2})I_{p,q,i,j-1/2}] + \\ \frac{F_{i,j}}{r_{i}w_{p,q}} [2\alpha_{p,q+1/2}I_{p,q,i,j} - (\alpha_{p,q+1/2}+\alpha_{p,q-1/2})I_{p,q-1/2,i,j}] + \\ \left[\frac{\hat{n} \cdot \Omega_{p,q}}{r_{i}\Delta\tau\Delta z} r_{i,j}^{w} L_{i,j}^{w} \right] I_{p,q,i,j} + (\kappa + \sigma)F_{i,j}I_{p,q,i,j} &= F_{i,j}S_{i,j} + S_{p,q,i,j}^{w} (19) \end{split}$$

which can be solved for $I_{p,q,i,j}$ and limited appropriately.

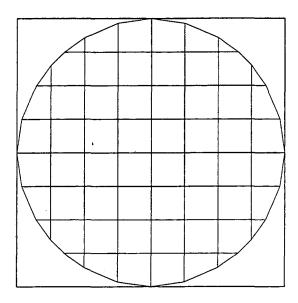


Figure 1: A circular enclosure represented as an embedded boundary in a uniform 8 × 8 grid.

As in the previous section, we update the boundary source term $S_{p,q,i,j}^w$ after every ordinate sweep so that the reflection source will always be up-to-date. The general form of the boundary condition follows the same pattern as for Cartesian coordinates.

Numerical Examples

The first two examples in this section establish second-order convergence of the heat transfer to the walls for straight and curved geometries, with black or gray walls. The final example shows a more complicated geometry representing an axisymmetric furnace. All ordinate values are taken from the S_6 set listed in [11], which has correct half-range first moments in the directions parallel to the coordinate axes.

Though we have written our code in a vectorizable form so that it will couple efficiently with fluid dynamics simulations running on Cray computers, all numerical examples presented here were computed on a DEC Alpha.

Scaled by π	Scaled by Length	Scaled by Area

Mesh	Absorption	Rate	Mesh	Absorption	Rate	Mesh	Absorption	Rate
8	0.8087779775		8	0.8148179684		8	0.8249516079	
16	0.8152651038	2.363	16	0.8167151729	3.221	16	0.8193067597	1.681
32	0.8165258819	3.523	32	0.8169186113	0.110	32	0.8175457223	1.432
64	0.8166355411	1.185	64	0.8167301495	0.668	64	0.8168932723	1.437
128	0.8165873005	0.413	128	0.8166114962	1.125	128	0.8166522731	1.506
256	0.8165510810	1.178	256	0.8165571095	1.407	256	0.8165674273	1.587
512	0.8165350736	1.346	512	0.8165365979	1.463	512	0.8165391793	1.591
1024	0.8165287755	1.435	1024	0.8165291575	1.508	1024	0.8165298054	1.596
2048	0.8165264457	1.505	2048	0.8165265420	1.550	2048	0.8165267038	1.612
4096	0.8165256248		4096	0.8165256490		4096	0.8165256894	

Table 1: Scaled heat transfer to the wall of a black circular enclosure of unit diameter. Mesh sizes range from 8×8 to 4096×4096 .

Black Circular Enclosure

We first consider the circular enclosure of unit diameter shown in Figure 1. The walls are cold and black, and the interior is a uniform absorbing medium with blackbody emissive power $\pi I_b = 1$ and $\kappa = 2$. Table 1 gives the average heat transfer per unit length along the embedded boundary, which is the only boundary in the problem. Each value r_i given for convergence rate is a function of the three numbers a_{i-1} , a_i , a_{i+1} in the sequence to its left. We define

$$r_i = \log_2 \left| \frac{a_i - a_{i-1}}{a_{i+1} - a_i} \right|, \tag{20}$$

so that r_i will indicate the order of accuracy of the scheme: 1 for first-order, 2 for second-order and so on. We calculate the average flux through the boundary as follows:

$$\operatorname{Flux} = \frac{1}{L} \sum_{i,j} (1 - \rho) \sum_{\hat{n} \cdot \Omega_m > 0} (\hat{n} \cdot \Omega_m) w_m L_{i,j}^w I_{m,i,j}^w. \tag{21}$$

The L in this equation represents the total length of the circular boundary, but there is some question of how best to define this for the discrete problem at hand. For the first column of the table we use $L=\pi$, the length of the circle itself. For the second column we take L equal to the length of the polygonal representation of the circle on the discrete mesh. The errors are slightly greater for this case, but the convergence rate is more consistent. Finally, for the third column we take L=2A/r, with r=0.5 and A being the discrete area of the region. This gives the most consistent convergence rates—even for coarser grids—suggesting that this interpretation comes closest to capturing the discrete behavior of the system. (To see why scaling by area is reasonable, consider that since the algorithm is conservative, the heat transferred to the walls is equivalent to the difference between emission and absorption in the interior medium.)

All three columns are asymptotically equivalent, since $L \to \pi$ and $A \to \pi/4$, both to second order. The overall convergence rate for the scheme appears to be a bit less than second order. In the next section we will see that a problem with straight sides comes much closer to second-order convergence. We speculate that the degraded performance for curved boundaries is an effect of the changing surface normals as the mesh is refined, interacting with the fixed ordinate directions, so that $\sum_{\hat{n}\cdot\Omega_m>0}(\hat{n}\cdot\Omega_m)w_m$ takes on completely different values for each mesh.

The results in Table 1 are presented to 10 decimal places only for the purpose of showing the convergence behavior of the scheme. The numbers given are not in any sense "correct" to this level of accuracy, so it is worthwhile to estimate the actual sizes of the errors in the calculation:

The largest error is that due to the choice of ordinate discretization. The correct analytic value for this problem is 0.8143 [12]. The limiting value of 0.8165 seen in the table shows the error in angular discretization with this particular S_6 ordinate set. The error is similar in size to those computed in [13] for three different ordinate sets, using a fixed unstructured grid roughly comparable to our 16×16 example. It is apparent from the table that at this resolution the errors due to the spatial and angular discretizations are at about the same order of magnitude.

The ordinates and weights are tabulated in [11] to seven decimal places. We rescale these so that $\sum_{m} w_{m}$ is exactly 4π , and in the boundary condition (9) we replace π with the actual half-range first moment in the manner of equation (16). With these changes the system is conservative to machine precision for black walls. For gray walls, as in the next example, the system is conservative to the accuracy to which we solve the reflecting boundary condition, about twelve decimal places.

The errors due to the spatial discretization are apparent from the table, being at about the seventh decimal place for the finest grids. Though the errors due to the choice of ordinate set are much larger, they should not affect the observed convergence rates since the ordinate set is the same for all of the runs.

Gray Axisymmetric Enclosure

For our second example we consider the geometry shown in Figure 2. The region is a unit square, except for the side on the right which slants inward at a 20° angle. We use axisymmetric coordinates with the left side falling on the axis, so the actual three-dimensional geometry forms a frustum of a cone. Only the slanting wall is modeled as an embedded boundary—the axis and the two remaining walls fall on the edges of the domain and are modeled as in a uniform mesh. Note that two cells contain both domain and embedded boundaries.

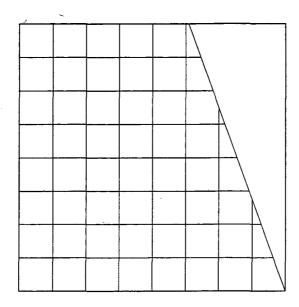


Figure 2: An axisymmetric enclosure represented using a combination of embedded and domain boundaries.

The interior is a uniform absorbing medium with blackbody emissive power $\pi I_b = 1$ and $\kappa = 2$. The walls are all gray with $\rho = \epsilon = 0.5$ and $\pi I_b = 0.5$. The RTE is solved by initializing the wall reflection sources to zero and iterating until the change in reflection source from one iteration to the next is reduced by a factor of 10^{-12} . For the present problem with immediate updates at each boundary we saw approximately one order of magnitude reduction per iteration.

Table 2 gives the average heat transfer per unit area along the boundaries. The first column shows absorption alone along the slanting wall, the second column shows absorption minus emission, while the third column shows this same quantity for all three walls combined. Because the walls are all straight, the surface areas and enclosed volume are always correct to machine precision. Since L is a constant the scaling issues of the previous example do not arise. The convergence rate is nearly 2, and is roughly the same for each column. Thus the method is second-order for straight sides, and performs correctly with emitting and reflecting walls and with interactions between domain and embedded boundaries.

Axisymmetric Furnace

For our final example, we plot contours of incident energy $G_{i,j} = \sum_{p,q} w_{p,q} I_{p,q,i,j}$ for the furnace geometry shown in Figure 3. Furnace dimensions, wall temperatures, emissivities and reflectivities are taken from [14], while the interior is taken to be a cold absorbing medium with $\kappa = 0.2 \text{m}^{-1}$. A 40×227 grid is used with

	Embedded Wall Absorption			Embedded Wall Net Flux			Total Wall Net Flux			
1	Mesh	Absorption	Rate	Mesh	Absorption	Rate	Mesh	Absorption	Rate	
	8	0.4603048202		8	0.2102362789		8	0.2085897529		
	16	0.4605075961	2.598	16	0.2101239715	0.005	16	0.2084476802	0.715	
	32	0.4604741120	-0.332	32	0.2100120832	0.864	32	0.2083611186	1.132	
1	64	0.4604319634	0.935	64	0.2099505878	1.198	64	0.2083216156	1.322	
	128	0.4604099129	1.444	128	0.2099237879	1.527	128	0.2083058114	1.555	
	256	0.4604018086	1.697	256	0.2099144862	1.733	256	0.2083004335	1.752	
	512	0.4603993098	1.839	512	0.2099116890	1.855	512	0.2082988363	1.871	
	1024	0.4603986113	1.933	1024	0.2099109159	1.939	1024	0.2082983996	1.931	
1	2048	0.4603984284	1.944	2048	0.2099107143	1.949	2048	0.2082982850	1.963	
<u></u>	4096	0.4603983809		4096	0.2099106621		4096	0.2082982556		

Table 2: Scaled heat transfer to the wall of an axisymmetric gray radiating enclosure.

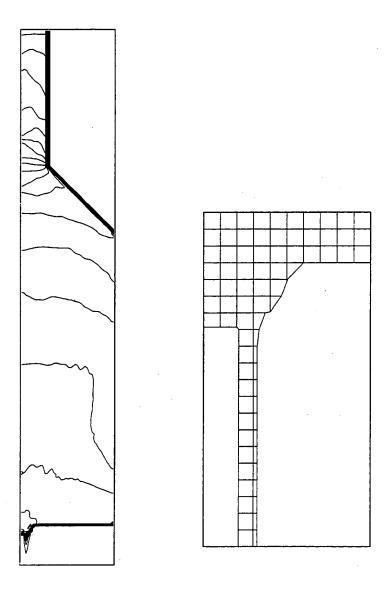


Figure 3: Left: Contours of incident energy in the BERL geometry. Right: Detail of grid around inlet and burner region at lower left.

axisymmetric coordinates. Ray effects are visible at the lower left due to the relatively cold inlet region, and in the exhaust duct due to the temperature differential between the duct and the furnace.

This example is provided to show the use of the embedded boundary algorithm on a more complex geometry. It is not a realistic model of radiation in the furnace since fluid emission is neglected; see [5] for a more complete simulation.

The geometry includes fine details of the burner region that are not well-represented on the discrete mesh. The inset shows how the grid in this region only roughly captures the shape of the burner—for details of the actual geometry see [14]. Accurate simulations will require a combination of the embedded boundary techniques presented here, with the adaptive mesh refinement algorithms presented in [15] and [3].

Conclusions

We have presented an extension of the discrete ordinates method for radiative transfer calculations to domains with embedded boundaries. An embedded boundary is represented as a piecewise-linear tracked front within a regular mesh using volume and area fractions for each cell. The method supports both Cartesian and axisymmetric meshes. Both embedded and domain-wall boundary conditions can coexist

within a single problem or even within a single cell.

The discretization is equivalent to second-order diamond differences in the mesh interior, and reduces to first-order at the boundaries. Flux limiting is implemented in a way that does not inhibit vectorization. The solution strategy is based on transport sweeps, iterated to convergence of the wall reflection and scattering terms. Due to the form of the reflection updates, energy is conserved to the level of the convergence tolerance. (With minor modifications the method could conserve to machine precision independent of this tolerance, but at the expense of slower convergence.)

Numerical examples illustrate the method for a curved embedded boundary in Cartesian coordinates, and for a combination of straight embedded and domain-wall boundaries in axisymmetric coordinates. Both black and gray walls are considered. The results show clear second-order convergence of the heat transfer to the walls for the straight wall example. The curved example shows a convergence rate intermediate between first and second order. A final example shows the method applied to a more complex geometry, that of an axisymmetric furnace.

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