

1 Rafting-Enabled Recovery Avoids Recrystallization in 3D-

2 Printing-Repaired Single-Crystal Superalloys

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18

19 **Abstract**

20 The repair of damaged Ni-based superalloy single-crystal turbine blades has been
21 a long-standing challenge. Additive manufacturing is a promising route to this end,
22 but a formidable obstacle is that recrystallization seems inevitable in the dislocation-
23 riddled heat-affected zone, bringing forth new grains that degrade the high-
24 temperature creep properties. Here we design a post-3D-printing recovery protocol
25 that eliminates the driving force for recrystallization, namely the stored energy
26 associated with the high dislocation content, prior to standard solution treatment and
27 aging. The post-electron-beam-melting, pre-solution recovery via sub-solvus annealing
28 is rendered possible by the rafting of γ' particles that facilitates dislocation
29 rearrangement and annihilation. The rafted microstructure is removed after solution
30 annealing, leaving behind a damage-free single crystal with uniform γ' precipitates
31 that is indistinguishable from the the rest of the turbine blade. This discovery offers a
32 practical means to keep 3D-printed parts from recrystallizing into a polycrystalline
33 microstructure, paving the way for additive manufacturing to repair, restore and
34 reshape any superalloy single crystal product.

35

36 Introduction

37 Modern turbine blades are made of superalloy single crystals (SXs), strengthened
38 by cuboidal precipitates of ordered γ' -phase in the γ -phase matrix ¹. SX Ni-based
39 superalloys outperform their polycrystalline counterparts by a large margin, in many
40 aspects including the resistance to creep, fatigue, and oxidation ², by eliminating most
41 of the defects that form during blade casting. These high-value SX blades are
42 nonetheless subject to surface damage and cracking upon extensive service in harsh
43 environments. It is thus critical to find a way to repair the damaged surface while
44 keeping their single crystalline nature as well as the desired uniform γ/γ'
45 microstructure. Such a successful repair will extend the life of turbine blades and
46 reduce the overall cost significantly.

47 The versatile 3D printing route emerged in recent years appears to be a powerful
48 option towards this goal ³. Via “epitaxial” deposition of the alloy, one layer at a time,
49 additive manufacturing can preserve the crystallographic orientation of the substrate
50 SX ⁴⁻¹¹. However, as 3D-printing involves fast cooling, the γ' precipitated are
51 excessively small with uneven sizes and rounded corners, and hence less stable during
52 high-temperature service. More importantly, a high density of dislocations build up in
53 the heat-affected zone (HAZ), due to unavoidable (often local) deformation under high
54 thermal stresses during printing ^{12,13}. Upon solution treatment at elevated temperatures,
55 these regions riddled with defects readily undergoes recrystallization (RX) ¹⁴ that
56 renders the microstructure polycrystalline, which would significantly degrade the high
57 temperature creep performance of blades (Figure S1 in Supporting Information) ².

58 These shortcomings are in fact characteristic of all 3D-printed superalloys ¹⁵, such that
59 the repaired part is no longer as strong as the original SX. There is thus a pressing
60 need to conceive an innovative treatment that can resolve this problem.

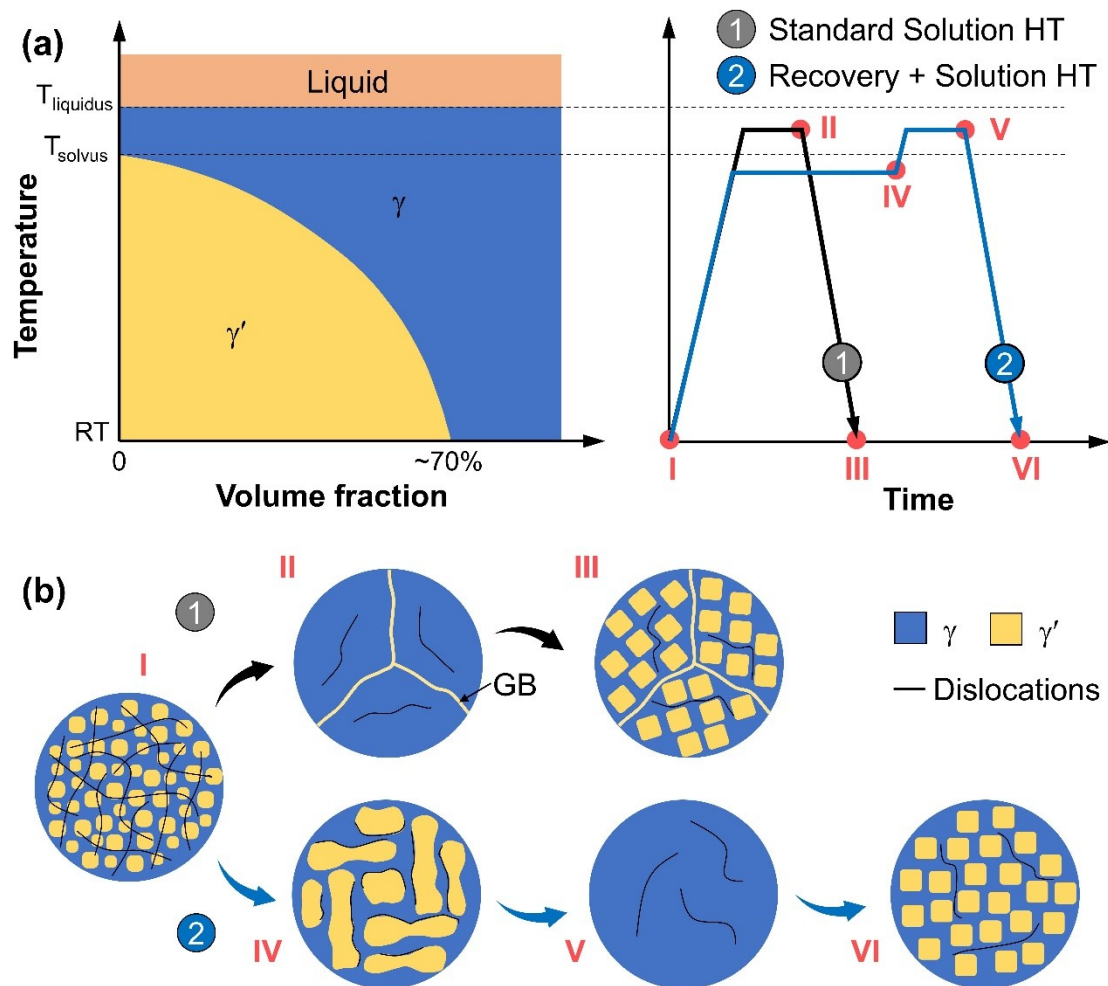
61 To this end, our strategy is to design post-3D-printing annealing to reduce the
62 driving force for recrystallization that ruins the SX structure. That is, we aim to
63 remove the accumulated dislocations through a recovery heat treatment (HT) before
64 the standard solution treatment and aging HT. However, conventional wisdom is that
65 recovery is difficult to realize in Ni-based superalloys for mainly two reasons. First,
66 the stacking fault energy of Ni-based superalloys ($< \sim 20 \text{ mJ/m}^2$) ^{16,17} is much lower
67 than that of pure Ni (125 mJ/m^2) ¹⁸ and Al (166 mJ/m^2) ¹⁹. A low stacking fault energy
68 promotes the unit dislocations to dissociate into partial dislocations, hampering their
69 climb and cross slip, the basic mechanisms for recovery ^{19,20}. Second, the closely
70 spaced γ' particles impede the motion of stored dislocations and hence prevent their
71 annihilation at temperatures below the γ' solvus ^{19,21,22}. If, instead, the alloy is heated to
72 temperatures above the γ' solvus, RX sets in quickly well before the stored defects get
73 effectively removed via recovery ²⁰⁻²³. Such an RX scenario is demonstrated in Figure
74 1 (standard solution HT).

75 In this paper, we demonstrate a novel HT process to produce a repaired single
76 crystal with indistinguishable γ/γ' microstructure from the interior. Prior to standard
77 solution annealing where all the γ' disappear, other defects (with associated excess
78 energy) accumulated in the HAZ can already be programmed to be relieved by taking
79 advantage of rafting-facilitated recovery, through an HT step of annealing at an

80 appropriate sub-solvus temperature. During such recovery annealing prior to solution
 81 HT, rafted microstructure forms in the HAZ, opening pathways to dislocation
 82 rearrangement and annihilation. This greatly reduces the dislocation density, such that
 83 in subsequent solution and aging HT, using standard protocol normally used, RX does
 84 not get triggered to nucleate new grains.

85

86 Results



87

88 Figure 1 Microstructural evolution of 3D printed Ni-based single crystals induced by standard solution
 89 HT, in comparison with our novel HT incorporating recovery annealing. (a) Standard solution HT
 90 involves one-step annealing between the solvus and liquidus temperatures, while our new HT protocol
 91 includes a recovery annealing procedure prior to solution HT. (b) RX is triggered by the standard

92 solution HT, but preempted by the new HT.
93

94 Electron-beam melting (EBM) with no powder feeding was carried out as an
95 analogue to 3D-printing repairing on AM3, a first-generation SX Ni-based superalloy
96 used as a model in this study. The electron beam was focused and programmed for line
97 scans on the substrate (base metal, BM) surface of [001] cast SX boule to generate
98 fusion zones (FZs). Figure 2 shows the morphology of the EBM sample and the
99 crystal orientation distribution before and after HT using different protocols. In the
100 low-magnification image, Figure 2a, three regions can be readily distinguished: un-
101 affected BM, HAZ, and FZ. The crystal orientation is the same throughout the as-
102 prepared EBM sample, from the BM to the HAZ to the FZ (Figure 2b). After solution
103 HT at 1300 °C for 30 min without prior recovery annealing, RX grains and high-angle
104 grain boundaries are clearly observed in both HAZ and FZ (Figure 2c). In contrast,
105 after our new 1100 °C recovery annealing for 6 h prior to standard HT, the electron
106 backscatter diffraction (EBSD) map shows no detectable RX grains (Figure 2d). In
107 other words, the newly developed HT protocol fulfills the “keeping the single crystal”
108 requirement. The other requirement that needs to be accomplished simultaneously is
109 the “uniform γ' microstructure”.

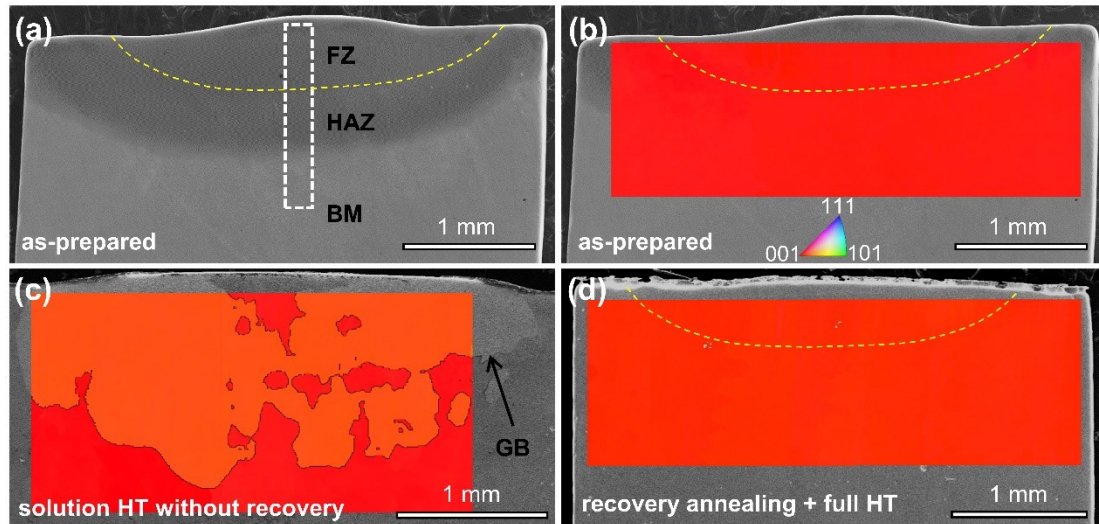


Figure 2 EBM samples before and after different HT protocols. (a) Three regions are visible in the cross-sectional scanning electron microscope (SEM) image of the as-prepared sample: fusion zone (FZ), heat-affected zone (HAZ), and unaffected base metal (BM). EBSD orientation maps indicate that (b) the as-prepared EBM sample is a single crystal, but (c) RX occurs and high-angle grain boundaries are generated upon solution HT at 1300 °C for 0.5 h, while (d) the RX is successfully prevented to keep the single crystalline nature by applying a recovery HT at 1100 °C for 6 h before standard HT.

The EBM sample (Figure 2b) without HT does not fit the bill in this regard. There the fast cooling rate experienced produces not only much finer dendrites in the FZ compared to the cast counterpart, but also much smaller γ' precipitates - the average size is < 50 nm (Figure 3a), only a quarter of that in the BM. A microstructure gradient is observed in the HAZ (in between the FZ and the BM). The upper HAZ near the fusion line has experienced a temperature sufficiently high to dissolve all the primary γ' . Thus the small particles in the upper HAZ are re-precipitated γ' , quite similar to those in the FZ. Moving down away from the fusion line, the peak temperature is lower and the period of time that the local temperature is above the γ' solvus is shorter, such that the primary γ' only dissolves partially. The subsequent re-precipitation at lowered temperatures forms fine secondary γ' . Together they constitute

a bimodal γ' size distribution. The population of the secondary γ' decreases gradually with increasing distance, becoming completely undetectable in the region about 300 μm away from the fusion line, leaving only cuboidal primary γ' with side length in the 200 – 300 nm range.

After EBM, the crystal is riddled with dislocations and the elastic strain associated with them is distributed in an inhomogeneous manner. To see the latter, we have taken synchrotron X-ray microdiffraction (μXRD) across the FZ and the HAZ, covering an area of 80 μm (horizontal) $\times$ 1150 μm (vertical), as indicated using the boxed rectangle in Figure 2a, at 2 μm spatial resolution. From the orientation inverse pole figure (IPF) maps (Figure S2 in Supporting Information), it appears that the entire scanned area remains a single crystal and the dendrite growth is along the [001] direction. The vertical and horizontal components of the elastic strain tensor, denoted as ϵ_v and ϵ_h , respectively, are displayed in Figure 3b. In all regions from the BM through the HAZ to the FZ, these strains are found to vary considerably in magnitude from location to location, and even change sign. The BM substrate is under tension in the vertical direction but is compressed in the horizontal direction; this is believed to be related to the pre-printing thermal history of the superalloy. The opposite is observed for strains in the HAZ: ϵ_h is tensile while ϵ_v is compressive. The transition from the BM to the HAZ is smooth and gradual. Both ϵ_h and ϵ_v change sign near the BM/HAZ interface. ϵ_h and ϵ_v reach their peak magnitudes at the HAZ/FZ interface, and then decrease together upon entering the FZ. The magnitude of the strain is highly variable in the FZ, and changes sign from the interdendritic regions to dendrite cores,

probably due to non-uniform chemical distribution. A high density of dislocations in the HAZ is reflected by the obviously-broadened Laue peak width, see the colored peak width map in Figure 3d (although the dislocation density is difficult to quantify accurately from the peak width). In the FZ, the peak width map also exhibits fine stripe features, presumably attributable to the chemical and microstructural heterogeneities between dendrite cores and interdendritic regions. This evidence of a large population of dislocations in many local regions is consistent with the observation in the transmission electron microscope (TEM) image of Figure 3e, in which the dislocation density is measured to be about $8 \times 10^{14} \text{ m}^{-2}$. This value is for a small local region, but from the colored peak width map the dislocation density is inhomogeneous and should be higher in many other regions of the HAZ. Similar to previous reports for cast materials ²⁴, the BM is almost defect free (Figure 3f).

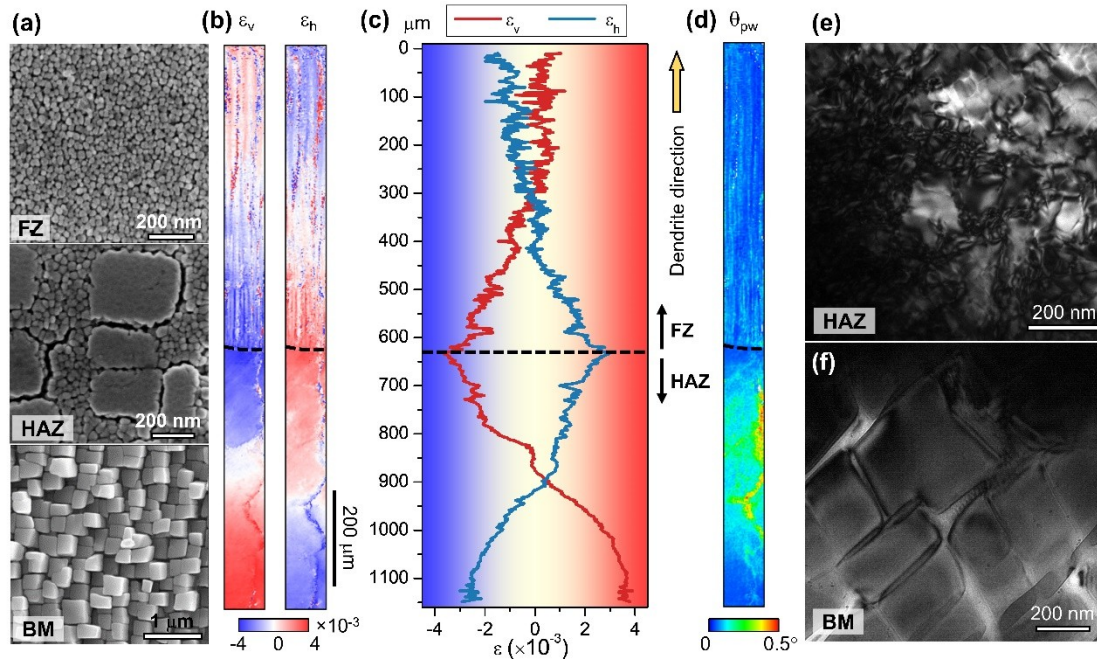


Figure 3 Inhomogeneous distributions of γ' morphology, elastic strain and dislocations in the EBM sample. (a) Distinctly different γ' morphologies in the FZ, HAZ, and BM, as displayed in their corresponding SEM images. (b) 2D and (c) 1D lattice elastic strain as well as (d) diffraction peak width

maps indicate non-uniform deformation. The yellow-red region in (d) is believed to represent a dislocation density of $> 8 \times 10^{14} \text{ m}^{-2}$. The fusion line is marked using a dashed line in (b-d). TEM images indicate that dislocations of high density are present in the HAZ (e), while the BM is almost defect free (f).

We stress again that the dislocations and elastic energy stored are the root cause that drives RX upon the ensuing solutionizing treatment. After the EBM specimen was solution treated at 1300 °C for 30 min, EBSD found RX grains in not only the HAZ but also the FZ (Figure 2b). Shortening of the solution treatment time reduced the area fraction of the RX grains, but was not able to avoid RX, unless the solution treatment is conducted at a temperature well below that needed to reach complete solid solution. Such a temperature, however, would not be able to solutionize the interdendritic region, where the elements promoting γ' formation are known to be enriched and thus the γ' solvus temperature is higher than that in the dendrite cores. In other words, without the proposed pre-solution recovery annealing, either RX occurs (at super-solvus temperatures) or the solution treatment is ineffective (at sub-solvus temperatures) to remove the chemical and microstructural heterogeneities from dendrites.

After incorporating our recovery annealing before solution HT, the microstructure becomes completely different from not only the reference sample (full HT after EBM without recovery annealing; as seen in Figure 2c, there RX is obvious, although the precipitates inside the grains may become uniform in size), but also the EBM sample (non-uniform microstructure, as discussed above with Figure 3). The γ' precipitates in the FZ, HAZ, and BM are uniform, exhibiting identical morphology and size. As seen

191 in Figure 4a, they all have cubical shape with sharp vertices, straight edges, and
192 uniform side length of about 500 nm. The γ' precipitates grow during the aging HT,
193 becoming larger than those in the BM of the EBM sample. From the micro-
194 indentation test results shown in Figure S3, after HT with recovery the hardness is also
195 uniform throughout, all the way from the FZ to the HAZ and to the BM, in stark
196 contrast to the non-uniform hardness distribution in the EBM sample due to the
197 pronounced spatial variation of the γ' precipitates size and morphology (Figure 3a) as
198 well as of the dislocation density. μ XRD results prove that the residual strain is fully
199 released, evidenced by the uniform light color in the 2D maps, Figure 4b. There is
200 only slight variation in the 1D strain profile of Figure 4c. The dislocation density in
201 the HAZ and the FZ is also brought down significantly. The TEM image in Figure 4e
202 shows a dislocation density of $2 \times 10^{13} \text{ m}^{-2}$. Multiple TEM images are taken from the
203 HAZ, two of which are displayed in Figure S4. From these different regions the
204 average dislocation density in the HAZ is found to be about $3 \times 10^{13} \text{ m}^{-2}$, and in all
205 local regions it never exceeds $5 \times 10^{13} \text{ m}^{-2}$. In other words, compared to the as-
206 prepared EBM sample, the dislocation density decreases by more than 20 times in the
207 HAZ. The BM, similar to the as-prepared sample, stays dislocation free (Figure 4f).

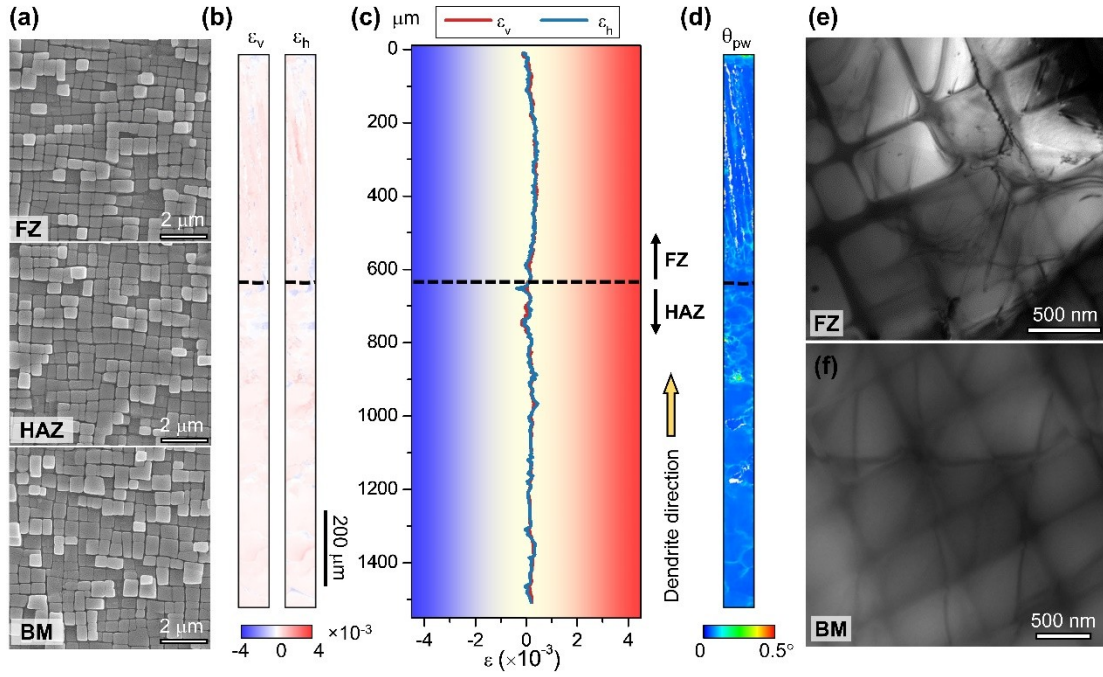


Figure 4 Uniform distributions of γ' morphology, elastic strain and dislocations after full HT following our new protocol incorporating recovery annealing. (a) Indistinguishable SEM images of γ' precipitates in the FZ, HAZ, and BM. μXRD proves (b, c) completely relieved strain, and (d) uniform and sharp diffraction peaks indicating low defect density (except at dendrite boundaries). The fusion line is marked with a dashed line in (b-d). TEM images indicate significant reduction in dislocation density in the HAZ (e), while BM remains almost defect free (f).

All the evidences above, including the γ' shape, size, and micro-hardness distributions, as well as the reduced strain and dislocation density, demonstrate that our new protocol is effective in producing a uniform microstructure. The key step, absent for the standard HT, is the sub-solvus annealing at 1100 °C for 6 h before solution treatment. It lowers the defect density that would have otherwise triggered RX during solution HT. Our discussion in the following will therefore focus on the mechanism involved to accomplish dislocation removal during this important recovery annealing step.

As shown in Figure 5, recovery annealing at 1100 °C for 6 h leads to rafting

almost everywhere in the FZ and the HAZ. Rafting is a common occurrence well known in superalloys subjected to external stresses and elevated temperatures. It can also be induced by residual stresses at certain temperatures²⁵. In our case the rafted microstructure is caused by the residual stresses built-in during the EBM. The recovery annealing is carried out at a temperature close to the typical creep testing temperature for this superalloy²⁶. Three typical rafted γ' microstructures (demonstrated in Figure 5a), which are recorded in the upper FZ, near the FZ/HAZ interface, and in the lower HAZ, respectively, are examined in detail. For AM3, a superalloy with negative lattice mismatch, the rafting direction is expected to be perpendicular to the tensile stress direction. In the upper FZ region, both horizontally and vertically rafted structures are observed, agreeing well with the observed spatial variation of the strain tensor and the sign (stress direction) displayed in Figure 3. Near the interface between the FZ and HAZ, vertically rafted structure forms. This is also consistent with the measured horizontal tensile strain in this region. As the strain direction reverses sign in the lower HAZ, the rafting direction changes as well. In confirmation, the strains in the recovery-annealed specimen are significantly reduced, suggesting that the elastic strain energy stored in the EBM sample is effectively released along with rafting when the γ/γ' phase boundaries migrate (Figure 5b, c). The peak width map indicates that there are residual defects in the HAZ and the lower FZ (Figure 5d), thus TEM is employed for direct observation. As shown in Figure 5e and 5f, dislocations now line up at γ/γ' phase boundaries, and dislocation networks are also observed occasionally. In the literature, these two types of dislocation

247 configurations were reported in superalloys after creep testing ^{27,28}. The dislocation
248 density near these interfaces is measured to be $9 \times 10^{13} \text{ m}^{-2}$ and $2 \times 10^{14} \text{ m}^{-2}$,
249 respectively, but still up to one order of magnitude lower than that in the HAZ of the
250 EBM sample. Taken together, the μ XRD and TEM observations suggest that the
251 combined rafting-recovery is akin to that due to creep ^{25,29,30}. Note here again that
252 previous annealing efforts were only able to achieve limited recovery, especially for
253 regions that experienced rather high pre-strains in plastic deformation ^{31,32}. Most of the
254 previous attempts apparently have missed the appropriate temperature window, as they
255 were probably unaware of the potential role that could be played by the rafting of γ'
256 particles.

257 Regarding rafting, it is in fact the migration of the γ/γ' phase boundaries that has
258 ushered in a new mechanism to facilitate dislocation recovery. First, many dislocations
259 sink into the interphase boundaries as they sweep by. Second, more spaces are opened
260 for dislocation motion, as rafting widens some γ channels to accommodate
261 dislocation movement and interactions that lead to annihilation. Third, the residual
262 dislocations rearrange into lower-energy configurations at the γ/γ' interfaces (Figure
263 5) and these aggregated dislocation bundles re-configure more readily upon
264 subsequent solution treatment at higher temperatures, as they no longer need to run
265 across γ' precipitates to annihilate. This reduction of concentrated defects storing high
266 energy leaves few spots as RX nucleation sites, such that the ensuing solution
267 treatment no longer sets off RX.

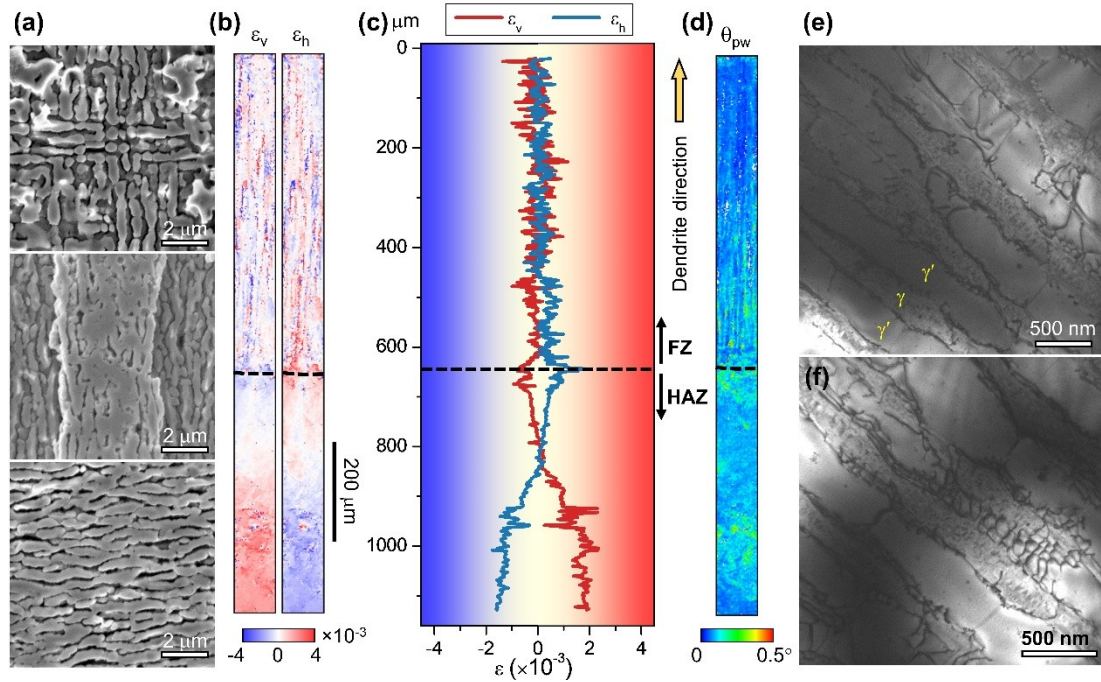


Figure 5 γ' morphology, elastic strain and dislocations after 1100 °C recovery annealing. (a) Rafted γ' observed in various directions. μ XRD shows lowered strain in (b) 2D and (c) 1D profiles, as well as (d) reduced diffraction peak width. The fusion line is marked by dashed line in (b-d). TEM observation illustrates dislocations rearranged (e) at γ/γ' interfaces and (f) as networks. These are presumably low-energy dislocation configurations that can readily annihilate upon solution treatment, and the dislocation densities in these two images are almost one order of magnitude lower than that in the HAZ of the EBM sample.

Discussion

The recovery HT protocol reported in this study is applicable to other 3D printing scenarios. To demonstrate this, epitaxial AM3 layers were also manufactured using the direct laser forming method on top of the cast BM with the same dimensions as those for the EBM case. The subsequent heat treatments are the same, incorporating the same recovery annealing. Homogeneous γ' precipitates are again obtained in all the FZ, HAZ, and BM regions, without any sign of RX, as shown with EBSD examination results in Figure S5 in Supporting Information.

Furthermore, although the recovery annealing costs 6 h, it is only a minor extension to the standard HT, which involves solution treatment at 1300 °C for 3 h followed by two steps of ageing at 1100 °C and 870 °C for 6 h and 20 h, respectively. The inserted recovery step is thus only a simple addition, with little additional cost while effectively curtailing the undesirable RX.

Finally, we mention in passing a tweak to the recovery approach to lower the defect density and hence the RX driving force: the substrate can be heated *in situ* along with the 3D printing (using electron beam, laser or other heating resource) process. This “preheating” may lower the temperature for recovery annealing and shorten its duration, depending on the preheating conditions. However, this substrate preheating is expected to demand a stringent temperature control: a substrate temperature too high would diminish the temperature gradient, jeopardizing the epitaxial deposition; A temperature too low, on the other hand, would still incur too much deformation in local regions and hence too many stored dislocations in the HAZ. As such one would still need to add rafting-assisted recovery to suppress RX altogether. Our results in this paper demonstrate that the delicate control of preheating is not necessary. As a limiting case, simply using post-3D-printing recovery annealing alone, with no preheating at all, is already adequate to get rid of the normally expected RX. The post-electron-beam-melting, pre-solution recovery parameters (annealing temperature and time) can be adjusted depending on the plastic strains that need to be recovered inside the printed/repaired alloy; a systematic documentation would however exceed the space limit of this Letter.

307

308 **Conclusions**

309 In summary, we have designed a new heat treatment protocol to satisfy the
310 requirement of “no RX together with uniform γ ”, mandated for 3D-printing repair of
311 Ni-based superalloy single crystals. Most essential in our strategy is the realization of
312 sub-solvus recovery prior to solution treatment, eliminating most of the stored defect
313 energy that drives nucleation of new crystals during solution treatment. This is made
314 possible by dislocation rearrangement and annihilation that are otherwise inactive in
315 the absence of rafting γ' . Previous 3D-printing work to produce single crystalline
316 superalloys was not aware of, and did not take advantage of, this vehicle that enables
317 substantial recovery. Our finding thus opens an avenue to make additive manufacturing
318 a widely applicable tool when dealing with the manufacture and repair of single-
319 crystal superalloy part. We envision this could be used for the welding of single-
320 crystal parts as well.

321

322 **Methods**

323 EBM with no powder feeding was carried out using a DMAMS Zcomplex3™
324 electron-beam 3D-printing system operated in 10^{-3} mbar vacuum. The substrate (BM)
325 was cut into a cylinder 13 mm in diameter and 4 mm in height, from [001] cast SX
326 boules after solid solution heat treatment. Electron beam of 15 mA was accelerated to
327 60 keV and focused onto the BM surface to form a melt pool. Line scanning was
328 programmed at the velocity of 10 to 15 mm/s to ensure epitaxial dendrite growth in

the melt pool. A fusion zone (FZ) of about 1500 μm in width and 800 μm in depth was generated. The EBM sample was then recovery-annealed at 1100 $^{\circ}\text{C}$ for 6 h, solution-treated at 1300 $^{\circ}\text{C}$ for 0.5 h, and then aged at 1100 $^{\circ}\text{C}$ and 870 $^{\circ}\text{C}$ for 6 h and 20 h, respectively. Note that although the solution treatment temperature was the same as the standard HT protocol, the duration needed was significantly shorter, because solute segregation in the EBM sample is much less than in its cast counterpart. Comparisons are made with an identical EBM sample heat treated without recovery, skipping the 1100 $^{\circ}\text{C}$ annealing step. This is the standard HT sample serving as the reference.

Micro-hardness test was carried out using a Vickers hardness indenter under the force control mode, after EBM as well as after recovery annealing. On each specimen, a matrix of indentations covered the area from the BM to the FZ. Each indent was at least 4 μm deep to exclude the surface effect, and neighboring indents were 105 μm apart to make sure the hardness value was not influenced by the plastic deformation around adjacent indents.

Before and after recovery-HT, the microstructure was examined under secondary electron mode in a SEM after etching in 25% phosphoric acid water solution at the voltage of 5 V for 10 s. RX was monitored by mapping the crystal orientation of the sample surface using EBSD, after electrochemical polishing in 10% perchloric acid alcohol solution at the voltage of 20~30 V for about 60 s. μXRD sample was electro-polished the same way, and then scanned using micro-focused synchrotron polychromatic X-ray beam at the Advanced Light Source of Lawrence Berkeley

National Laboratory³³. The collected Laue diffraction data were processed using a custom-developed software based on the peak position comparison method to measure the strain tensor distribution accurately³⁴. The defect density maps were obtained semi-quantitatively by plotting the Laue peak width distribution³⁵. TEM specimens were prepared using the conventional twin-jet electropolishing.

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Author contributions

K.C., E.M., Z.W.S. and J.L. designed the project. R.H. conducted the EBM and direct laser forming experiments. R.H. and Y.L. developed the HT protocol for the EBM specimens. S.L. performed the HT on the direct laser forming sample. W.Z. carried out TEM and μ XRD characterizations, and then analyzed and interpreted the μ XRD

372 data with R.H. under the supervision of K.C. and N.T. The paper was written by K.C.,
373 R.H., E.M., Z.W.S. and J.L. All authors contributed to discussions of the results.

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